

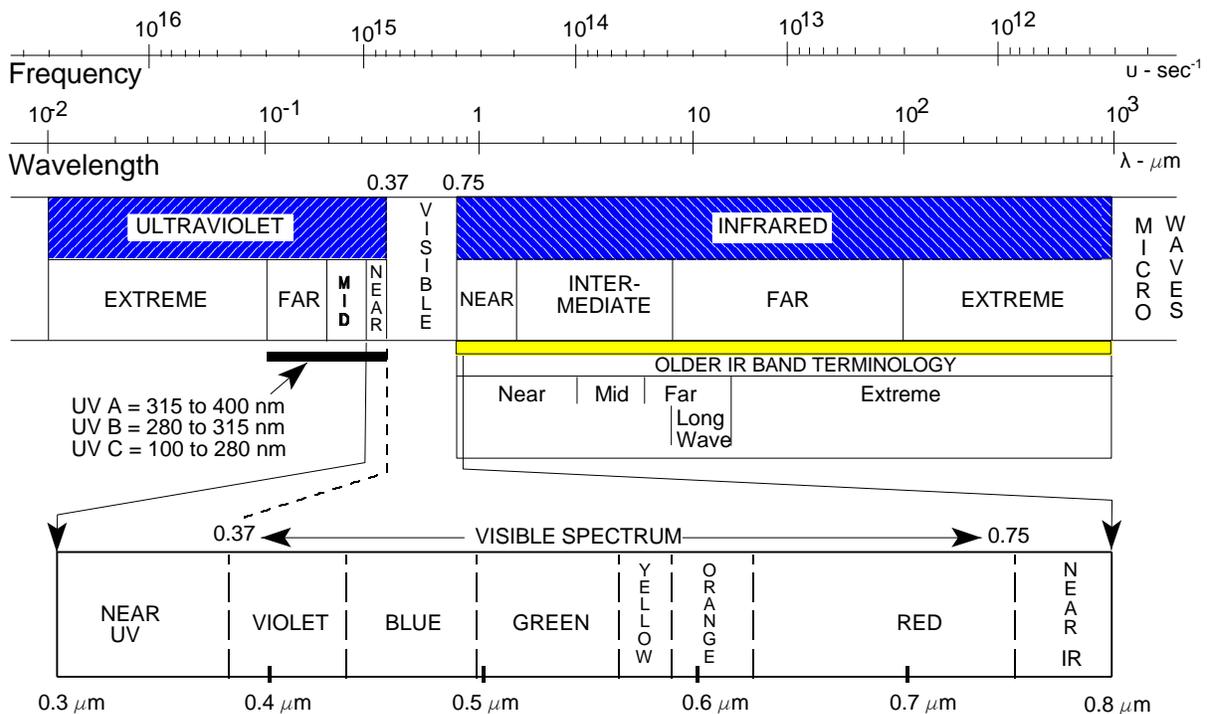
# ELECTRO-OPTICS

## INTRODUCTION

There are many electro-optical (EO) electronic warfare (EW) systems which are analogous to radio frequency (RF) EW systems. These EO EW systems operate in the optical portion of the electromagnetic spectrum. Electro-optics (EO), as the name implies, is a combination of electronics and optics. By one definition EO is the science and technology of the generation, modulation, detection and measurement, or display of optical radiation by electrical means. Most infrared (IR) sensors, for example, are EO systems. In the popularly used term "EO/IR," the EO is typically used to mean visible or laser systems. The use of EO in this context is a misnomer. Actually, almost all "EO/IR" systems are EO systems as defined above. Another often used misnomer is referring to an EO spectrum. EO systems operate in the optical spectrum, which is from 0.01 to 1000 micrometers. EO includes lasers, photometry, infrared, and other types of visible, and UV imaging systems.

## OPTICAL SPECTRUM

The optical spectrum is that portion of the electromagnetic spectrum from the extreme ultraviolet (UV) through the visible to the extreme IR (between 0.01 and 1000 micrometers ( $\mu\text{m}$ )). Figure 1 shows the optical spectrum in detail. Figure 2 shows the entire spectrum. The end points of the optical spectrum are somewhat arbitrary. On the long wavelength end of the spectrum IR radiation and microwaves overlap. Similarly, x-rays and the extreme UV overlap on the short wavelength end of the spectrum. How the division is made depends on one's point of reference. For example, radiation having a wavelength of 1000  $\mu\text{m}$  which is emitted from a very hot body and is detected by an energy measuring device such as a super-cooled bolometer is called IR radiation. However, radiation of the same wavelength (or 300 gigahertz) which is generated by an electric discharge and is detected by a bolometer in a waveguide is called microwave radiation. Older texts may refer to the terms near, middle, far, and far-far IR, the frequency limits of which differ from the newer divisions shown below. Notice that the preferred terminology no longer uses the term "middle IR".



**Figure 1. Optical Spectrum**

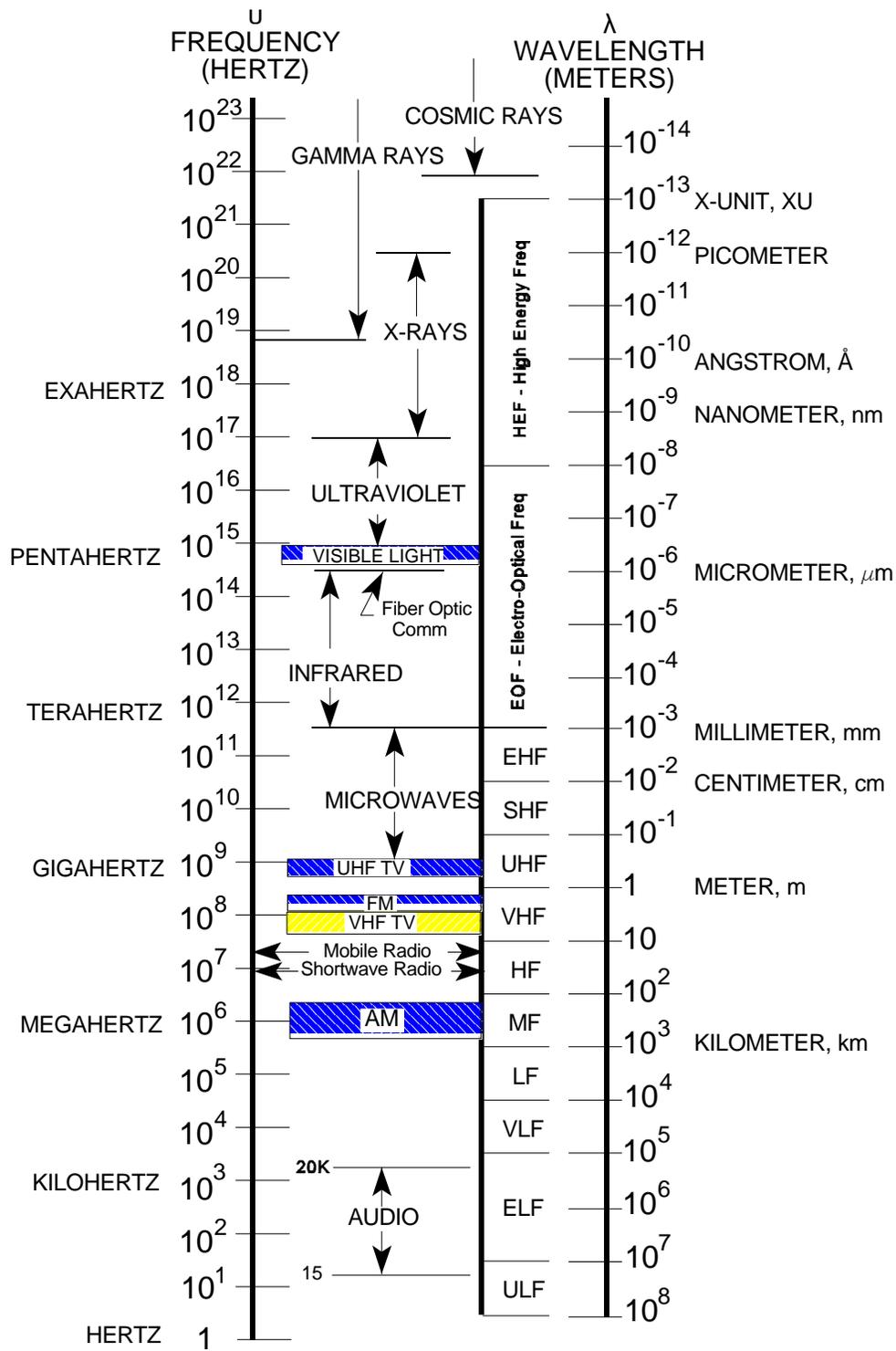


Figure 2. Electromagnetic Radiation Spectrum

## TERMINOLOGY

The common terms used to describe optical radiation are the source parameters of power, radiant emittance (older term) or radiant exitance (newer term), radiance, and radiant intensity. They refer to how much radiation is given off by a body. The parameter measured by the detector (or collecting object/surface) is the irradiance. Any of these quantities can be expressed per unit wavelength in which case the subscript is changed from e (meaning energy derived units) to  $\lambda$  and the term is then called "Spectral ...X...", i.e.  $I_e$  is radiant intensity, while  $I_\lambda$  is spectral radiant intensity. These quantities in terms of currently preferred "Système International d'Unités" (SI units) are defined in Table 1.

**Table 1.** Radiometric SI Units.

Symbol	Name	Description	Units
Q	Radiant Energy		J (joules)
$\Phi_e$	Radiant Power (or flux)	Rate of transfer of radiant energy	W (watts)
$M_e$	Radiant Exitance	Radiant power per unit area emitted from a surface	$W m^{-2}$
$L_e$	Radiance	Radiant power per unit solid angle per unit projected area	$W m^{-2}sr^{-1}$
$I_e$	Radiant Intensity	Radiant power per unit solid angle from a point source	$W sr^{-1}$
$E_e$	Irradiance	Radiant power per unit area incident upon a surface	$W m^{-2}$
$X_\lambda$	Spectral ...X..	(Quantity) per unit wavelength interval	(Units) $nm^{-1}$ or $\mu m^{-1}$
Where $X_\lambda$ is generalized for each unit on a per wavelength basis; for example, $L_\lambda$ would be called "spectral radiance" instead of radiance.			

In common usage, irradiance is expressed in units of watts per square centimeter and wavelengths are in  $\mu m$  instead of nanometers (nm). These previously accepted units and the formerly used symbols are known as the Working Group on Infrared Background (WGIRB) units, and are shown in Table 2. The radiant intensity is in watts per steradian in both systems.

**Table 2.** Older WGIRB Radiometric Units.

Symbol	Name	Description	Units
$\Omega$	Solid Angle		SR
$\lambda$	Wavelength		$\mu m$
P	Radiant Power	Rate of transfer of radiant energy	W
W	Radiant Emittance	Radiant power per unit area emitted from a surface	$W cm^{-2}$
N	Radiance	Radiant power per unit solid angle per unit projected area	$W cm^{-2}sr^{-1}$
J	Radiant Intensity	Radiant power per unit solid angle from a point source	$W sr^{-1}$
H	Irradiance	Radiant power per unit area incident upon a surface	$W cm^{-2}$
$X_\lambda$	Spectral ...X...	(Quantity) per unit wavelength	(Units) $\mu m^{-1}$

Other radiometric definitions are shown in Table 3.

**Table 3. Other Radiometric Definitions**

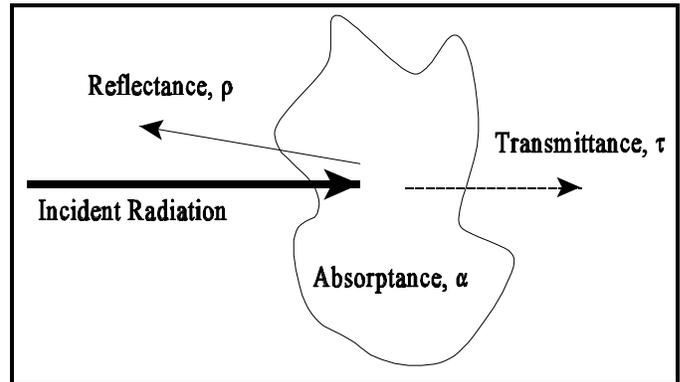
Symbol	Name	Description	Units
$\alpha$	Absorptance <sup>1</sup>	$\alpha = (*) \text{ absorbed} / (*) \text{ incident}$	numeric
$\rho$	Reflectance	$\rho = (*) \text{ reflected} / (*) \text{ incident}$	numeric
$\tau$	Transmittance	$\tau = (*) \text{ transmitted} / (*) \text{ incident}$	numeric
	Emissivity	$= (*) \text{ of specimen} /$ $(*) \text{ of blackbody @ same temperature}$	numeric

Where (\*) represents the appropriate quantity Q,  $\Phi$ , M, E, or L

Note (1) Radiant absorptance should not be confused with absorption coefficient.

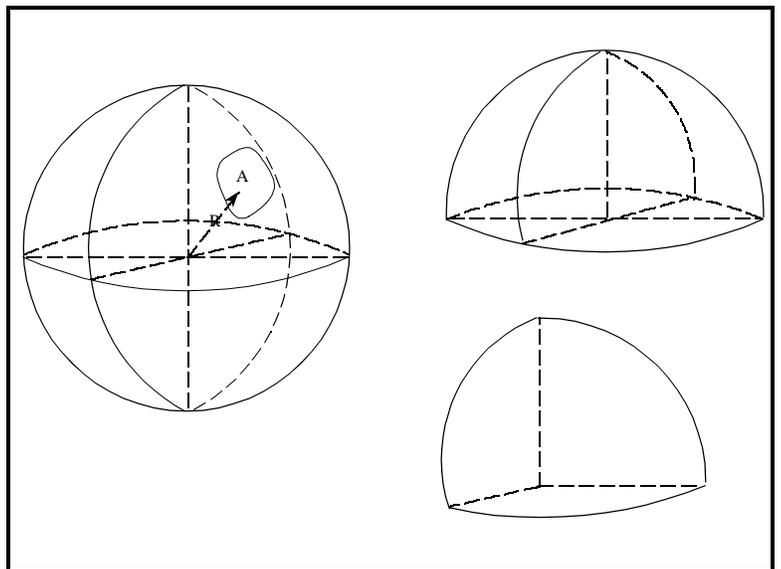
The processes of absorption, reflection (including scattering), and transmission account for all incident radiation in any particular situation, and the total must add up to one:  $a + \rho + \tau = 1$ , as shown in Figure 3.

A few words may be needed about the unit of solid angle, the steradian. Occasionally this unit is confusing when it is first encountered. This confusion may be partly due to difficulty in visualization and partly due to steradian being apparently a dimensionless unit (which is in itself a contradiction). Three solid angles are easy to visualize - these are the sphere, the hemisphere, and the corner of a cube (see Figure 4). There are  $4\pi$  steradians surrounding the center of a sphere,  $2\pi$  steradians in a hemisphere, and  $\frac{1}{2}\pi$  steradians in the corner of a cube (that is, the solid angle subtended by two walls and the floor of a room is  $\frac{1}{2}\pi$  steradians).



**Figure 3. Radiation Incident on a Body**

The problem of dimensions enters in calculating the steradiancy of a given area on a spherical surface. The number of steradians intercepted by an area A on the surface of a sphere of radius R is  $A/R^2$ . If length is measured in centimeters, the dimensions of the solid angle is  $\text{cm}^2/\text{cm}^2$ . So, steradian appears to be dimensionless. However, it is the unit, steradian, that is dimensionless (in terms of units of length), not the solid angle itself. One steradian is the solid angle intercepted by an area of one square centimeter on a spherical surface of one centimeter radius (or one square foot at one foot).



**Figure 4. Steradian Visualization**

IR wavelengths are typically expressed in  $\mu\text{m}$ , visible wavelengths in  $\mu\text{m}$  or  $\text{nm}$ , and UV wavelengths in  $\text{nm}$  or angstroms. Table 4 lists conversion factors for converting from one unit of wavelength to another. The conversion is from column to row. For example, to convert from  $\mu\text{m}$  to  $\text{nm}$ , multiply the value expressed in  $\mu\text{m}$  by  $10^3$ . IR wavelengths are also sometimes expressed in a frequency-like unit called wavenumbers or inverse centimeters. A wavenumber value can be found by dividing 10,000 by the wavelength expressed in  $\mu\text{m}$ . For example,  $2.5 \mu\text{m}$  converts to a wavenumber of 4000 or 4000 inverse centimeters ( $\text{cm}^{-1}$ ).

**Table 4. Wavelength Conversion Units**

From ->	Angstroms - $\text{\AA}$	Nanometers - $\text{nm}$	Micrometers - $\mu\text{m}$
To get	Multiply by		
Angstroms - $\text{\AA}$	1	10	$10^4$
Nanometers - $\text{nm}$	$10^{-1}$	1	$10^3$
Micrometers - $\mu\text{m}$	$10^{-4}$	$10^{-3}$	1

**PHOTOMETRIC QUANTITIES**

Whereas the radiometric quantities  $\Phi_e, M_e, I_e, L_e,$  and  $E_e$  have meaning throughout the entire electromagnetic spectrum, their photometric counterparts  $\Phi_v, M_v, I_v, L_v,$  and  $E_v$  are meaningful only in the visible spectrum ( $0.38 \mu\text{m}$  thru  $0.78 \mu\text{m}$ ).

The standard candle has been redefined as the new candle or candela ( $\text{cd}$ ). One candela is the luminous intensity of  $1/60$ th of  $1 \text{ cm}^2$  of the projected area of a blackbody radiator operating at the temperature of the solidification of platinum ( $2045 \text{ }^\circ\text{K}$ ). The candela (by definition) emits one lumen ( $\text{lm}$ ) per steradian.

Table 5 displays the photometric quantities and units. These are used in dealing with optical systems such as aircraft television camera systems, optical trackers, or video recording.

**Table 5. Photometric SI Units.**

Symbol	Name	Description	Units
$Q_v$	Luminous energy		lumen sec ( $\text{lm s}$ )
$\Phi_v$	Luminous flux	Rate of transfer of luminant energy	lumen
$M_v$	Luminous Excitance or flux density (formerly luminous emittance)	Luminant power per unit area	$\text{lm m}^{-2}$
$L_v$	Luminance (formerly brightness)	Luminous flux per unit solid angle per unit projected area	nit ( $\text{nt}$ ) or $\text{candela/m}^2$ or $\text{lm/sr m}^2$
$I_v$	Luminous Intensity (formerly candlepower)	Luminous power per unit solid angle from a point source	candela or $\text{lm/sr}$
$E_v$	Illuminance (formerly illumination)	Luminous power per unit area incident upon a surface	lux or $\text{lx}$ or $\text{lm/m}^2$
$K$	Luminous efficacy	$K = \Phi_v / \Phi_e$	$\text{lm / w}$

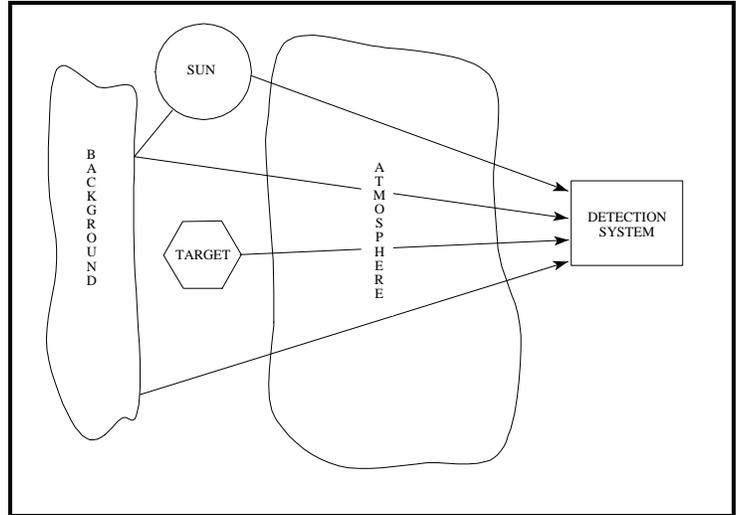
Table 6 displays conversion factors for commonly used illuminance quantities.

**Table 6.** Illuminance Conversion Units

		Lux (lx)	Footcandle (fc)	Phot (ph)
1 lux (lm m <sup>-2</sup> )	=	1	0.0929	1 x 10 <sup>-4</sup>
1 footcandle (lm ft <sup>-2</sup> )	=	10.764	1	0.001076
1 phot (lm cm <sup>-2</sup> )	=	1 x 10 <sup>4</sup>	929	1

**GENERALIZED DETECTION PROBLEM**

Figure 5 shows a generalized detection problem. On the left of the diagram are the radiation sources - the sun, background, and the target of interest. In the middle is the intervening atmosphere, which attenuates the radiation as it travels to the detection system shown on the right of the diagram.



**Figure 5.** Generalized Detection Problem

Anything at temperatures above absolute zero radiates energy in the electromagnetic spectrum. This radiation is a product of molecular motion, and the spectral distribution of the radiation is characterized by the temperature of the body. The four basic laws of IR radiation are Kirchoff's law, Planck's law, the Stefan-Boltzmann law, and Lambert's cosine law. Kirchoff found that a material that is a good absorber of radiation is also a good radiator. Kirchoff's law states that the ratio of radiated power and the absorption coefficient: (1) is the same for all radiators at that temperature, (2) is dependent on wavelength and temperature, and (3) is independent of the shape or material of the radiator. If a body absorbs all radiation falling upon it, it is said to be "black." For a blackbody the radiated power is equal to the absorbed power, and the emissivity (ratio of emitted power to absorbed power) equals one. One can also have a graybody - one which emits with the spectral distribution of a blackbody but at a lower intensity level because it has an emissivity of something less than one.

The radiation from a blackbody at a specific wavelength can be calculated from Planck's law:

$$W_{\lambda} = \frac{C_1}{\lambda^5 \left[ e^{\left( \frac{C_2}{\lambda T} \right)} - 1 \right]}$$

Where:  $C_1 = 2\pi^5 c^2 h / 15 = 3.7416 \times 10^{-12} \text{ W cm}^2$

$C_2 = ch/k = 1.4389 \text{ cm } ^\circ\text{K}$

$c = \text{speed of light}; h = \text{Planck's constant}; k = \text{Boltzman's constant}$

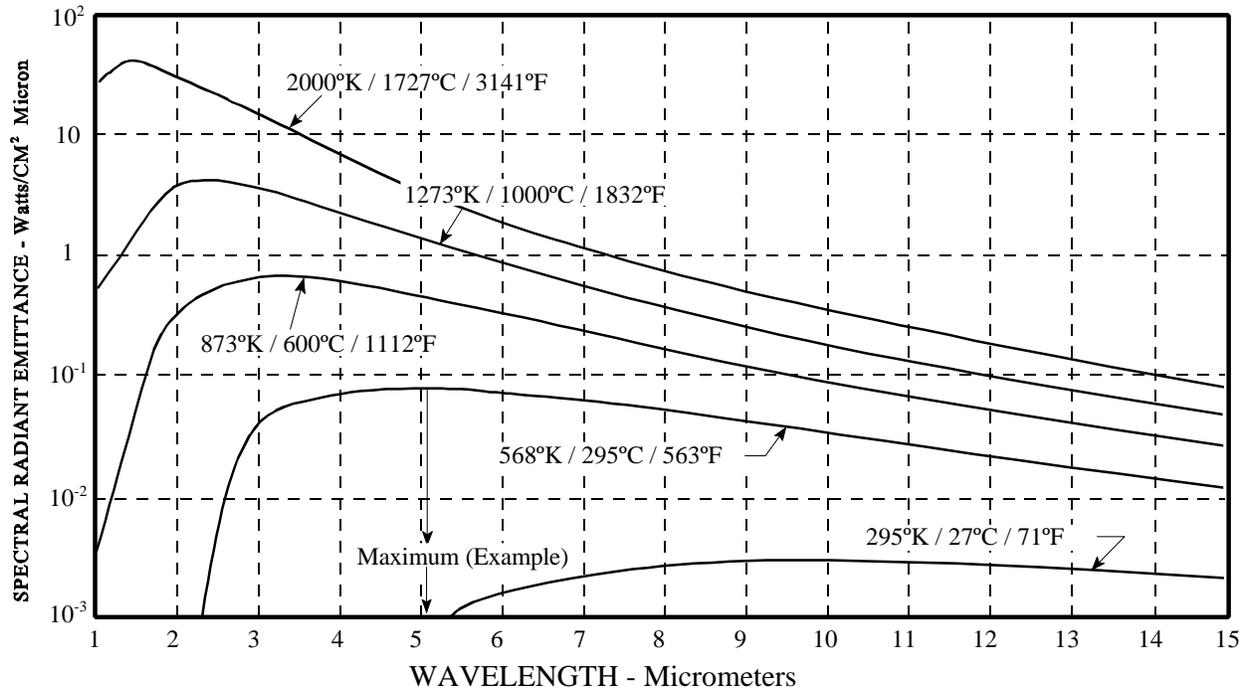
With  $\lambda$  in cm and T in  $^\circ\text{K} (= ^\circ\text{C} + 273)$

Figure 6 shows the spectral radiant emittance of blackbody radiators at several temperatures as calculated from this equation. [ $W_{\lambda}$  is in  $\text{W/cm}^3$  so multiply by  $10^{-4}$  to get  $\text{W/cm}^2\text{micron}$ ].

Wein's displacement law takes the derivative of the Planck's law equation (above) to find the wavelength for maximum spectral exitance (emittance) at any given temperature (or the temperature of maximum output at a given wavelength):

$$\lambda_m T = 2897.8 \mu^\circ\text{K}$$

For example, given that  $T=568^\circ\text{K}$ , then  $\lambda_m = 5.1\mu$  as verified by examining Figure 6.



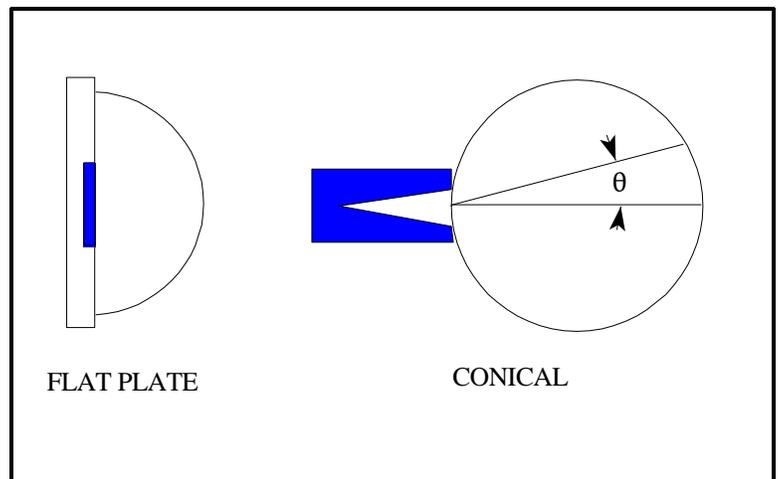
**Figure 6.** Blackbody Spectral Radiant Emittance

According to the Stefan-Boltzmann law, the total radiant emittance of a blackbody is proportional to the fourth power of the temperature:

$$W = \sigma T^4 \quad \text{Where: } \sigma = \frac{2\pi^5 k^4}{15c^2 h^3} = 5.67 \times 10^{-12} \text{Watts cm}^{-2} \text{ K}^{-4}$$

This is Planck's radiation law integrated over all values of  $\lambda$ .

A blackbody is a perfectly diffuse radiator. According to Lambert's law of cosines, the radiation emitted by a perfectly diffuse radiator varies as the cosine of the angle between the line of sight and the normal to the surface. As a consequence of Lambert's law, the radiance of a blackbody cavity is  $1/\pi$  times the radiant emittance (a conical blackbody cavity emits into a solid angle of  $\pi$  steradians). The radiation from a flat plate is emitted into  $2\pi$  steradians. The radiation pattern for these sources are shown in Figure 7. Notice that the conical cavity has the highest radiation straight ahead, and nothing at  $\theta$  angles approaching  $90^\circ$  whereas the flat plate has a uniform radiation pattern at all angles in front of the surface.



**Figure 7.** Blackbody Radiation Patterns

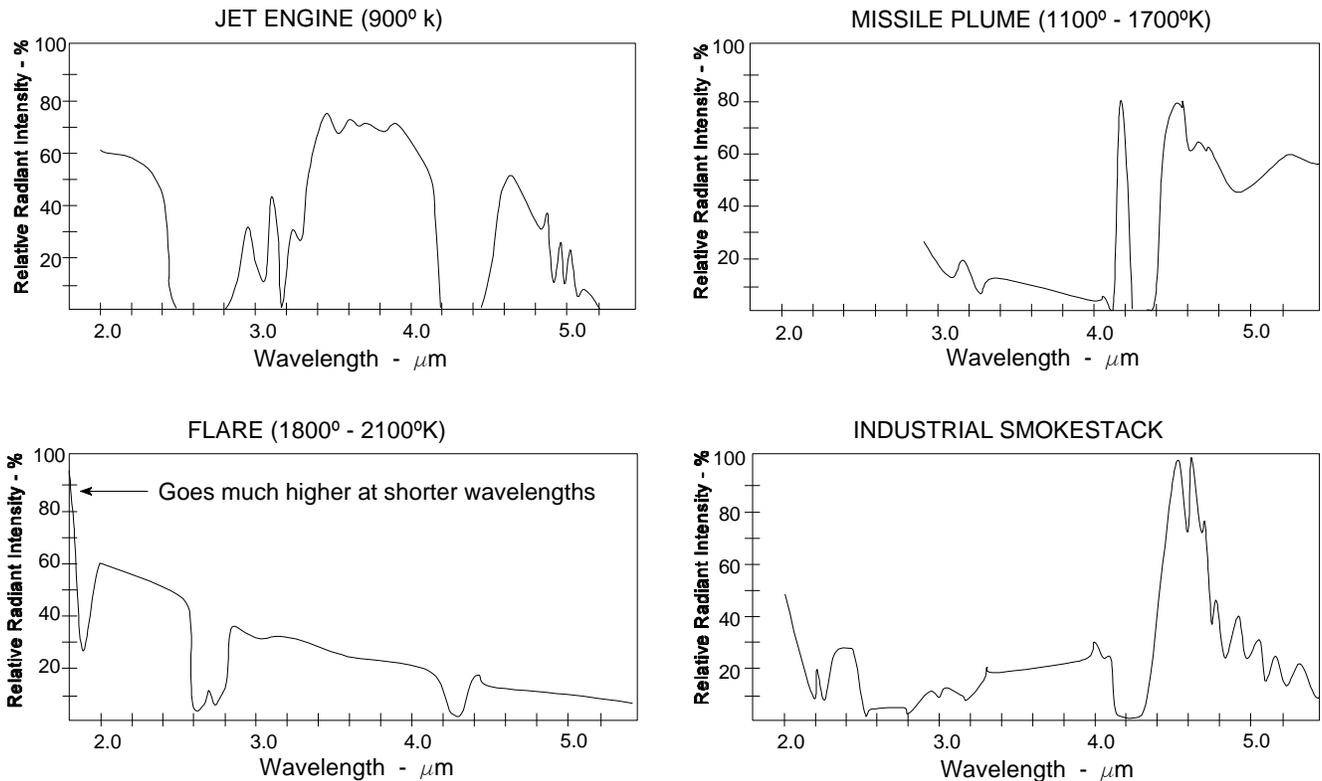
The interrelationship of the various quantities that describe source and received radiation in a vacuum are:

<u>SOURCE</u>		<u>RECEIVER</u>		
SI	WGIRB	SI	WGIRB	
$M_e = \Phi/A$	or $W = P/A$	$E_e = I_e/D^2$	or $H = J/D^2$	
$L_e = M_e/\pi$	or $N = W/\pi$			
$I_e = L_e A$	or $J = NA$			where A is the radiating area and D is the distance between source and receiver.

In actual practice the intervening atmosphere attenuates the radiation passing from the source to the receiver. When atmospheric transmission is accounted for, the receiver equation becomes:

$$E_e = \tau I_e / D^2 \quad \text{where } \tau \text{ is the atmospheric transmittance.}$$

The sources of radiation encountered outside the laboratory are either targets or backgrounds. One person's target may be another person's background. The target is the radiation source of interest - for example, an aircraft, a missile, a structure on the ground, or a ship at sea. The backgrounds are the non-target sources included within the field of view of the detection system which produce what amounts to noise - background noise. Possible background sources include the sun, clouds, terrain, the sea, blue sky, night sky, and stars. Figure 8 shows the spectral distribution of radiation from several targets and background sources. Spectral and spatial means are generally used to discriminate the target from the background. Spectral discrimination can be used because the targets are often characterized by spectral line or band emissions which yield a high signal to background ratio within a selected wavelength band. Also the target is usually small compared to the background so spatial discrimination can be used.



NOTE: These charts show relative not absolute radiant intensity of each signature. Consequently the "amplitude" of one cannot be compared with the "amplitude" of another.

**Figure 8.** Spectral Distribution of Various Targets

## ATMOSPHERIC TRANSMISSION

The radiation emitted or reflected from the targets and backgrounds must pass through the intervening atmosphere before reaching the detection system. The radiation is absorbed and re-emitted by molecular constituents of the atmosphere and scattered into and out of the path by various aerosol components. In the IR, atmospheric attenuation follows an exponential relationship expressed by the following equation:  $I = I_0^{-kD}$  where  $I_0$  is the radiation incident on the attenuating medium,  $k$  is the extinction coefficient, and  $D$  is the path length.

The molecules that account for most of the absorption in the IR region are water, carbon dioxide, nitrous oxide, ozone, carbon monoxide, and methane. Figure 9 shows the transmission of radiation over a 1 NM level path. The curve shows absorptions due to: 1) both water and carbon dioxide at 1.4  $\mu\text{m}$ , 1.85  $\mu\text{m}$ , and 2.7  $\mu\text{m}$ ; 2) due to water only at 6  $\mu\text{m}$ ; and 3) due to carbon dioxide only at 4.3  $\mu\text{m}$ .

Inspection of Figure 9 reveals the presence of atmospheric windows, i.e. regions of reduced atmospheric attenuation. IR detection systems are designed to operate in these windows. Combinations of detectors and spectral bandpass filters are selected to define the operating region to conform to a window to maximize performance and minimize background contributions. Figure 10 shows an expanded view of the infrared portion of the spectrum.

The transmission in a window is greatly dependent on the length and characteristics of the path. Figure 11 shows the transmission for a 15 NM path at 10,000-foot altitude with 100% relative humidity. As is readily apparent, the transmission in the windows is greatly reduced over the longer path compared to the transmission for the shorter path shown in Figure 9. Since water vapor generally decreases with altitude, transmission generally increases and path length becomes the determining factor. However, path length does not affect transmission of all wavelengths the same.

## ATTENUATION OF EM WAVES BY THE ATMOSPHERE

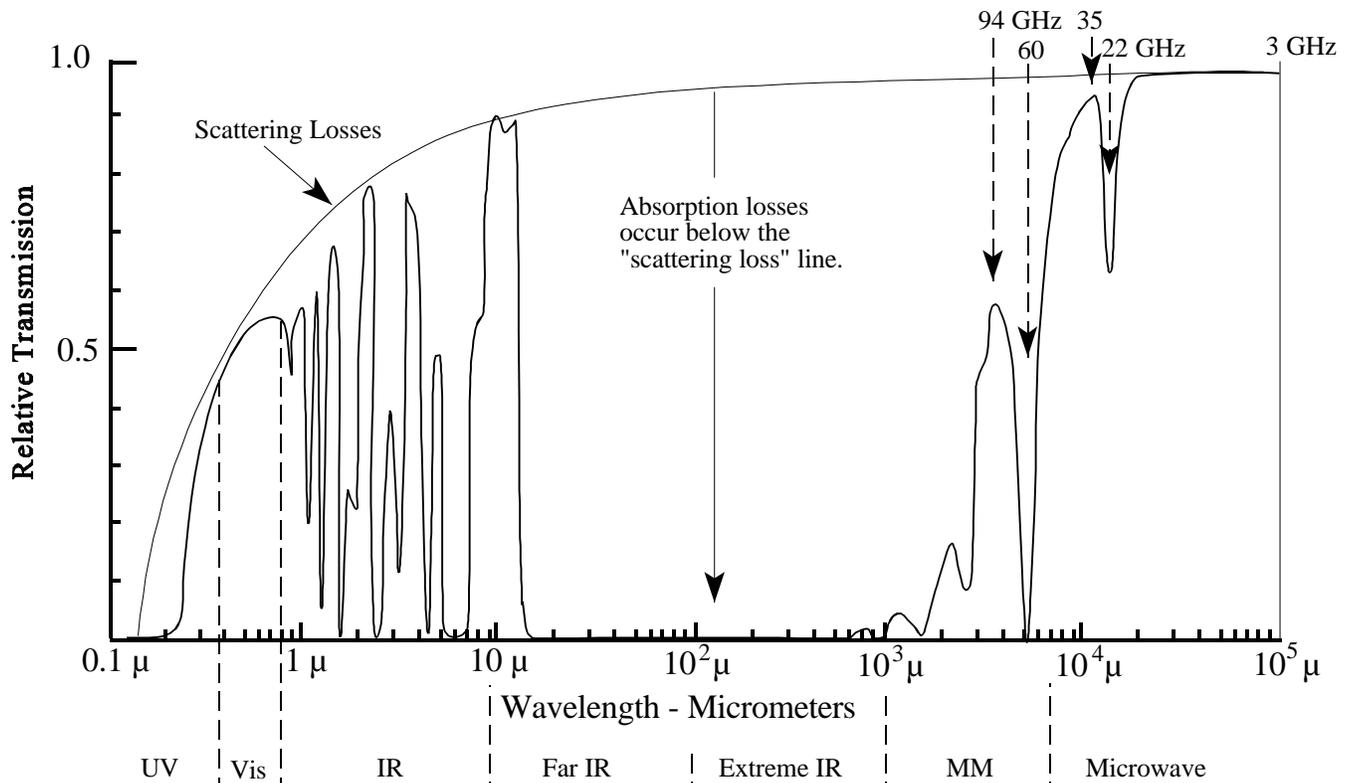
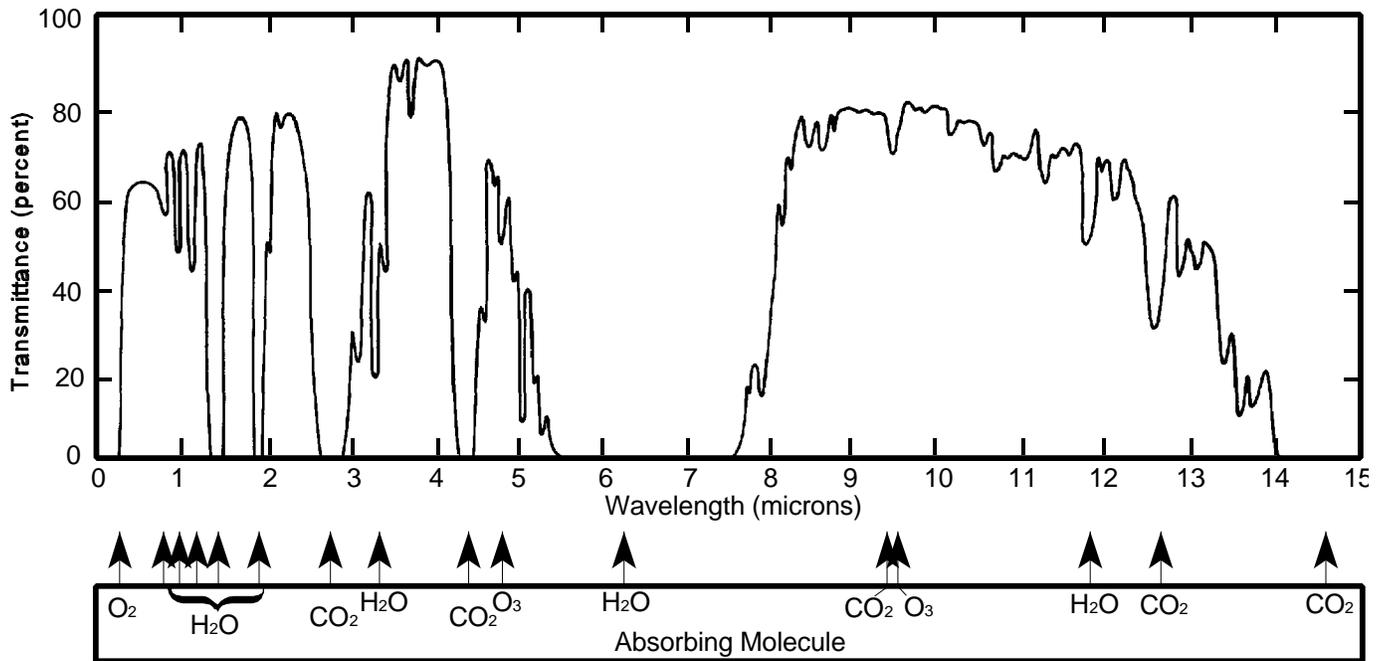
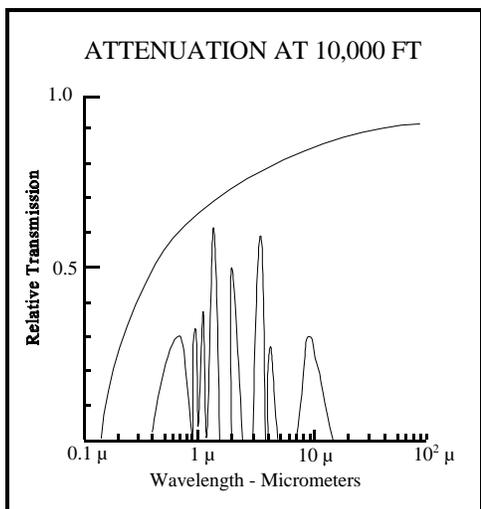


Figure 9. Atmospheric Transmission Over 1 NM Sea Level Path



**Figure 10.** Transmittance of Atmosphere Over 1 NM Sea Level Path (Infrared Region)



**Figure 11.** Atmospheric Transmission Over a 15 NM Path at 10,000 ft Altitude

## DETECTORS

A detector is a transducer which transforms electromagnetic radiation into a form which can be more easily detected. In the detectors of interest to EW the electromagnetic radiation is converted into an electrical signal. In some systems the signal is processed entirely within the system to perform its function. In others the signal is converted to a form to allow the human eye to be used for the final detection and signal analysis.

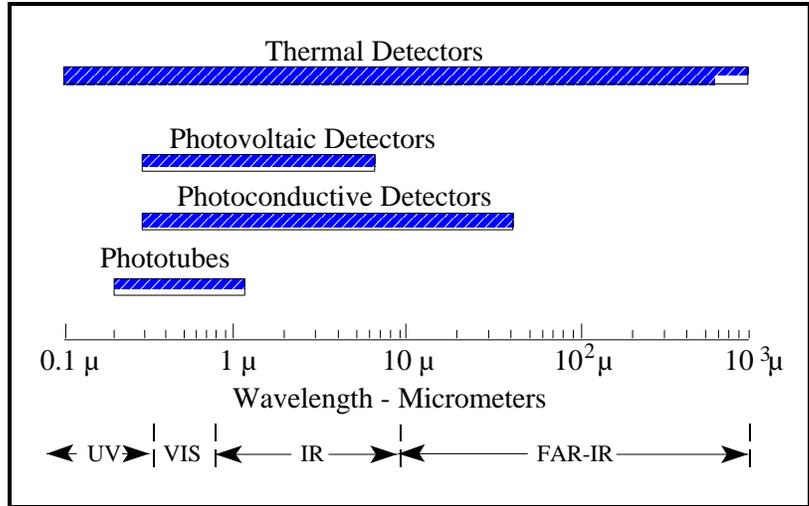
### Detection Mechanisms

The physical effects by which electromagnetic radiation is converted to electrical energy are divided into two categories: photon effects and thermal effects. EW systems primarily use detectors dependent on photon effects. These effects can be divided into internal photo effects and external photo effects. The external photo effect is known as photoemission. In the photoemissive effect, photons impinging on a photocathode drive electrons from its surface. These electrons may then be collected by an external electrode and the photocurrent thus obtained is a measure of the intensity of the received radiation.

Internal photoeffects of interest are the photoconductive effect and the photovoltaic effect. In the photoconductive effect, absorbed photons cause an increase in the conductivity of a semiconductor. The change is detected as a decrease in the resistance in an electrical circuit. In the photovoltaic effect, absorbed photons excite electrons to produce a small potential difference across a p-n junction in the semiconductor. The photovoltage thus produced may be amplified by suitable electronics and measured directly.

The pyroelectric effect is a thermal effect that is applicable to EW systems. The pyroelectric effect is a change in polarization in a crystal due to changes in temperature. Radiation falling on such a crystal is detected by observing the change in polarization as a build up of surface charge due to local heating. When coated with a good black absorber, the crystal will be sensitive to a wide band of wavelengths.

Figure 12 shows the spectral sensitivity range of typical detectors using these effects.

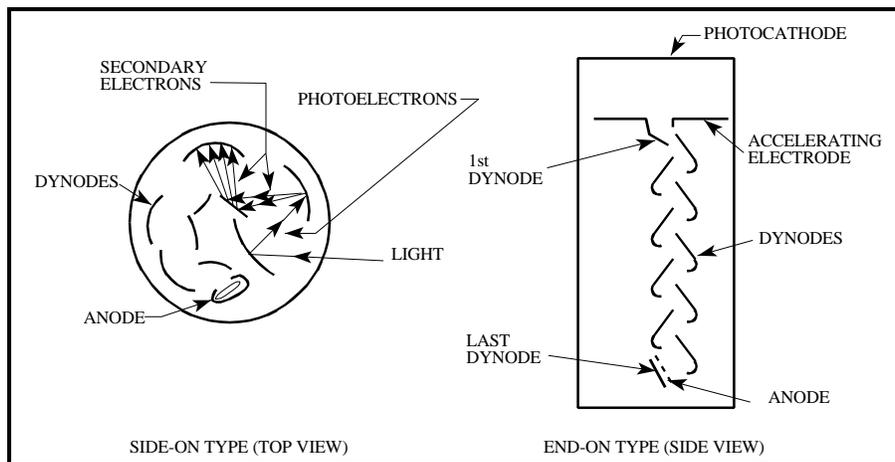


**Figure 12.** Spectral Range of Various Detectors

### Detector Types

Photon detectors exhibit sharp long wavelength cutoffs. The principle photoemissive detector type in EW systems is the photomultiplier. Current amplification is obtained in photomultipliers by secondary emission. A series of electrodes known as dynodes lie between the cathode and the anode. The structure of side-on and end-on type photomultipliers is shown in Figure 13.

The photoelectrons from the cathode are accelerated and focused onto the first dynode. Secondary electrons from the first dynode are accelerated and focused onto the second dynode, which emits more secondaries. This process is continued through from 4 to 16 stages in commercial tubes. Current gains of 10 million can be obtained with 16 stages. Typical response times (electron transit time) are tens of nanoseconds.



**Figure 13.** Multiplier Phototubes

Photoconductive detectors consist of a body of semiconductor - single or arrays- having electrodes attached to opposite ends. In operation they are used in electronic circuits as resistors whose resistance depends on the radiation upon the sensitive surface. Typical cooled and uncooled configurations are shown in Figure 14.

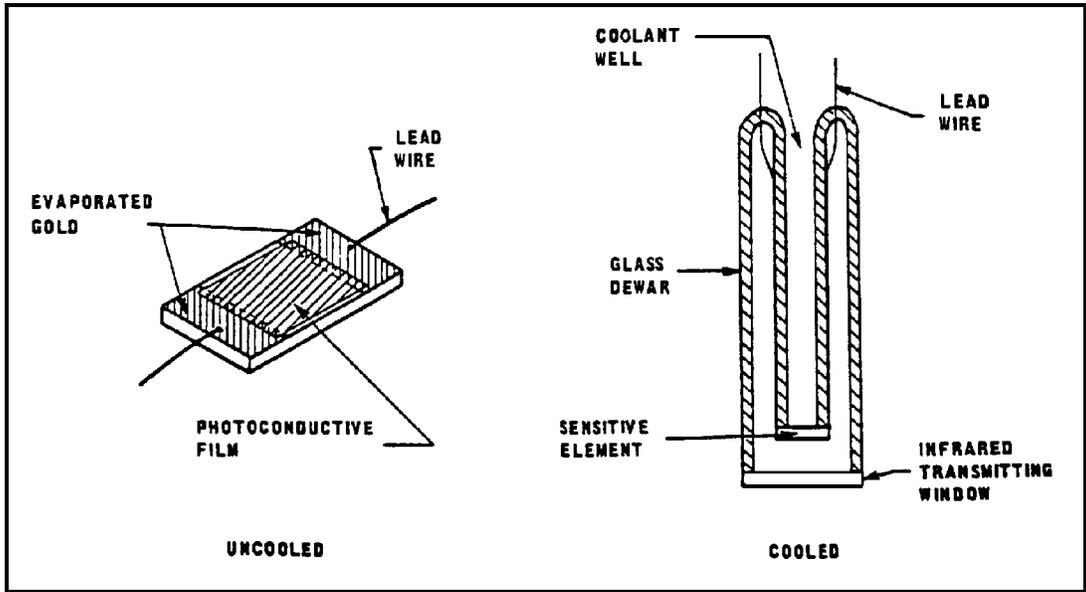


Figure 14. Photoconductive Detector

Photovoltaic detector configurations are shown in Figure 15. Photoconductive and photovoltaic detectors in EW systems are usually operated cooled for greater sensitivity. N-type material contains a large number of excess electrons and few “holes”, while P-type material contains few electrons and many holes.

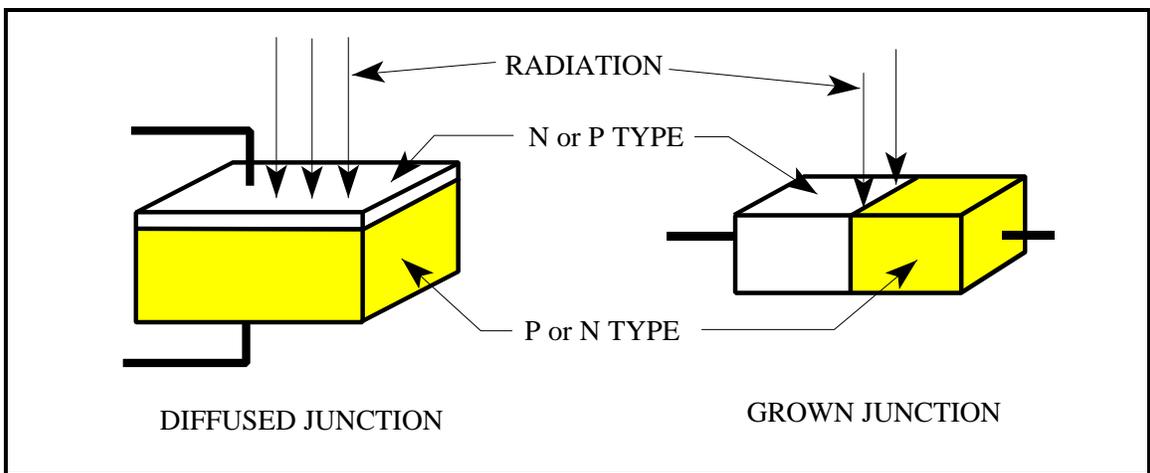


Figure 15. Photovoltaic Detector Configurations

Diode phototubes and photomultipliers are commonly used detectors for UV systems. The typical IR system uses arrays of photoconductive or photovoltaic detectors. Many state-of-the-art IR systems use what is known as focal plane arrays. The advantage of focal plane detectors is the ability to integrate processing electronics elements right on the same chip as the detector elements. Most visible band systems of interest are televisions. An example of a typical television camera tube is the vidicon (Figure 16). The vidicon is a storage type camera tube in which a charge-density pattern is formed by the imaged scene radiation on a photoconductive surface which is then scanned by a beam of low velocity electrons. The fluctuating voltage coupled out to a video amplifier can be used to reproduce the scene being imaged. Pyroelectric photocathodes can be used to produce a vidicon sensitive over a broad portion of the IR.

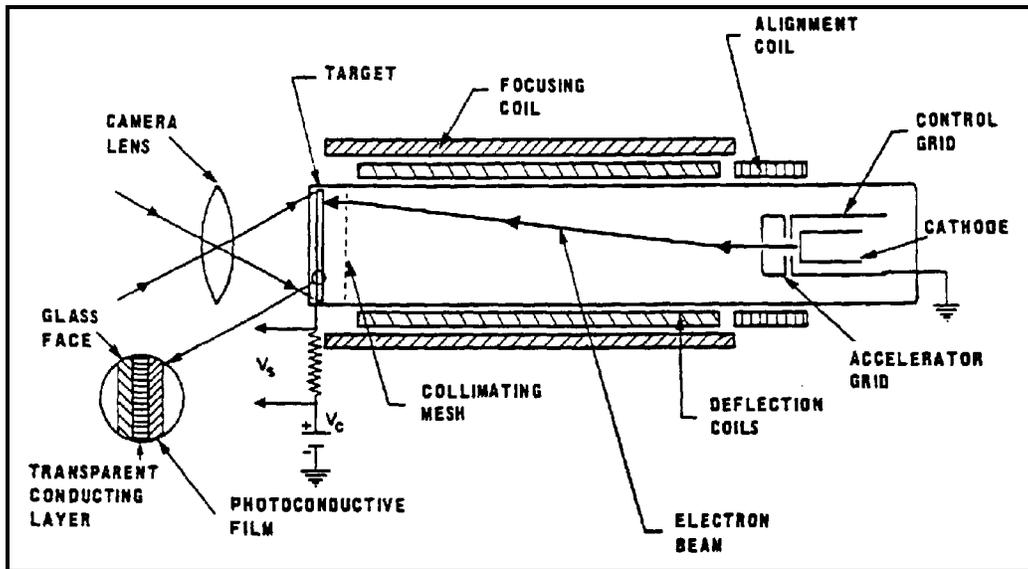


Figure 16. Vidicon

Another type of camera tube is the image orthicon which uses a photoemissive sensitive element (Figure 17). Small, light weight television cameras can now be made using charge-coupled device (CCD) or charge-injection device (CID) technology. CCD cameras are the basis of the popular hand-held camcorders.

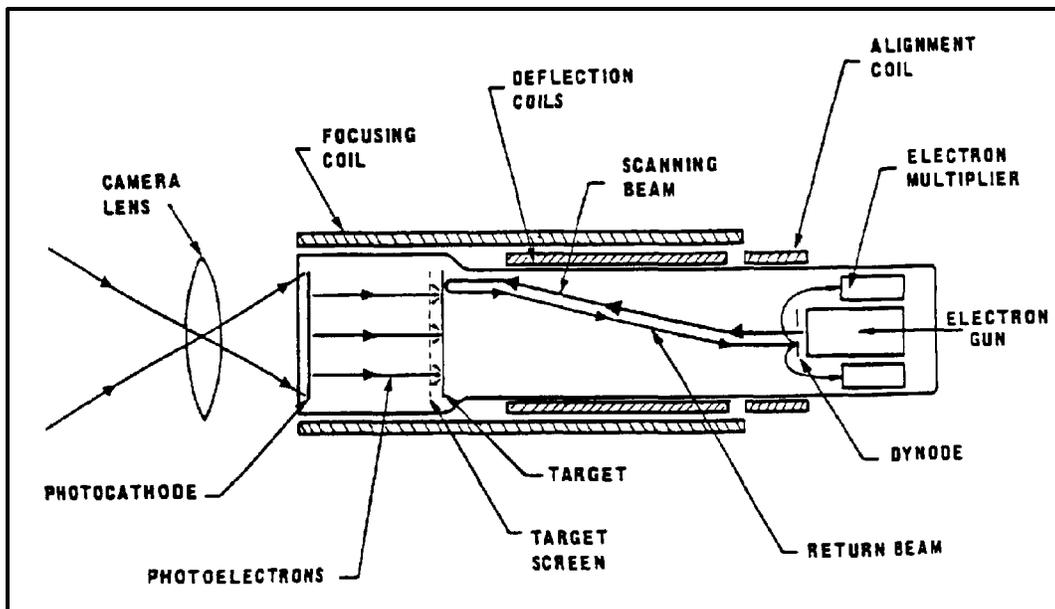


Figure 17. Image Orthicon

The most common detectors used in surface-to-air and air-to-air missile seekers use compounds which include:

Cadmium Sulfide	-	CdS	Lead Selenide	-	PbSe
Gallium Arsenide	-	GaAs	Lead Sulfide	-	PbS
Indium Antimonide	-	InSb			

Other known detector material includes:

Germanium doped with Copper	-	Ge:Cu	Germanium doped with Zinc	-	Ge:Zn
Germanium doped with Gold	-	Ge:Au	Indium Arsenide	-	InAs
Germanium doped with Mercury	-	Ge:Hg	Lead Telluride	-	PbTe
Mercury Cadmium Telluride	-	HgCdTe			

Some detectors (such as InSb) have multiple modes of operation, including: Photoconductive (PC), Photovoltaic (PV), or Photoelectromagnetic (PEM) modes of operation. Typical spectral detectivity characteristics for various detectors are shown in Figure 18.

### Detector Parameters and Figures of Merit

The important parameters in evaluating a detector are the spectral response, time constant, the sensitivity, and the noise figure. The spectral response determines the portion of the spectrum to which the detector is sensitive. The time constant is a measure of the speed of response of the detector. It is also indicative of the ability of the detector to respond to modulated radiation. When the modulation frequency is equal to one over the time constant, the response has fallen to 70.7 % of the maximum value. The time constant is related to the lifetime of free carriers in photoconductive and photovoltaic detectors and to the thermal coefficient of thermal detectors. The time constant in photoemissive devices is proportional to the transit time of photoelectrons between the photocathode and anode.

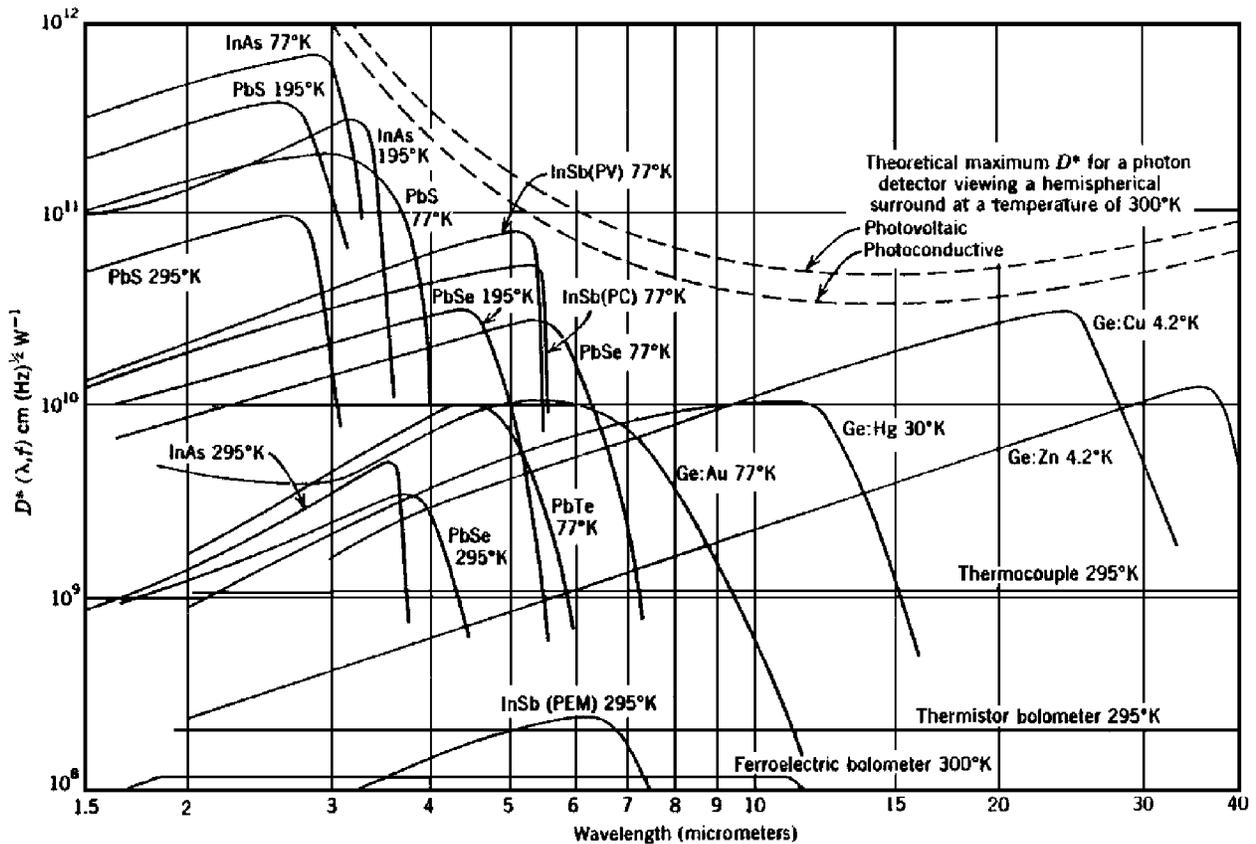


Figure 18. Spectral Detectivity of Various Detectors

The sensitivity of a detector is related to its responsivity. The responsivity is the ratio of the detected signal output to the radiant power input. For photoconductive and photovoltaic detectors the responsivity is usually measured in volts per watt -- more correctly, RMS volts per RMS watt. However, the sensitivity of a detector is limited by detector noise. Responsivity, by itself, is not a measure of sensitivity. Detector sensitivity is indicated by various figures of merit, which are analogous to the minimum detectable signal in radar. Such a quantity is the noise equivalent power (NEP). The NEP is a measure of the minimum power that can be detected. It is the incident power in unit bandwidth which will produce a signal voltage equal to the noise voltage. That is, it is the power required to produce a signal-to-noise ratio of one when detector noise is referred to unit bandwidth. The units of NEP are usually given as watts, but, more correctly, are  $\text{watts/Hz}^{1/2}$  or  $\text{watts}\cdot\text{sec}^{1/2}$ .

Another figure of merit is the noise equivalent input (NEI). The NEI is defined as the radiant power per unit area of the detector required to produce a signal-to-noise ratio of one. The NEI is obtained by dividing the NEP by the sensitive area of the detector. The units of NEI are watts per square centimeter. An NEI for photoemissive devices is commonly given in lumens.

The NEP has the disadvantage that better detectors have smaller NEP's, but the human psyche is such that a figure of merit that increases for improvements in detector performance is preferable. A figure of merit which has that feature is the detectivity (D), which is defined as the reciprocal of the NEP. The units of D are  $\text{watts}^{-1}\cdot\text{sec}^{-1/2}$ . A higher value of detectivity indicates an improvement in detection capability. The dependence on detector area is removed in another detectivity measure, known as D-star (D\*). D\* is the detectivity measured with a bandwidth of one hertz and reduced to a responsive area of one square centimeter. The units of D\* are  $\text{cm}\cdot\text{watts}^{-1}\cdot\text{sec}^{-1/2}$ . D\* is the detectivity usually given in detector specification sheets. The spectral detectivity is the parameter used in Figure 18.

Besides the NEI mentioned above, the quantum efficiency of the photocathode is also a figure of merit for photoemissive devices. Quantum efficiency is expressed as a percent -- the ratio of the number of photoelectrons emitted per quantum of received energy expressed as a percent. A quantum efficiency of 100 percent means that one photoelectron is emitted for each incident photon.

There are other figures of merit for television cameras. The picture resolution is usually described as the ability to distinguish parallel black and white lines and is expressed as the number of line pairs per millimeter or TV lines per picture height. The number of pixels in the scene also defines the quality of an image. A pixel, or picture element, is a spatial resolution element and is the smallest distinguishable and resolvable area in an image. CCD cameras with 512 x 512 elements are common. Another resolution quantity is the gray scale, which is the number of brightness levels between black and white a pixel can have.

### Noise in Detectors

The performance of a detector is limited by noise. The noise is the random currents and voltages which compete with or obscure the signal or information content of the radiation. Five types of noise are most prominent in detectors: thermal, temperature, shot, generation-recombination, and 1/f noise. Thermal noise, also known as Johnson noise or Nyquist noise, is electrical noise due to random motions of charge carriers in a resistive material. Temperature noise arises from radiative or conductive exchange between the detector and its surroundings, the noise being produced by fluctuations in the temperature of the surroundings. Temperature noise is prominent in thermal detectors. Shot noise occurs due to the discreteness of the electronic charge. In a photoemissive detector shot noise is due to thermionic emission from the photocathode. Shot noise also occurs in photodiodes and is due to fluctuations in the current through the junction. Generation-recombination noise is due to the random generation and recombination of charge carriers (holes and electrons) in semiconductors. When the fluctuations are caused by the random arrival of photons impinging upon the detector, it is called photon noise. When it is due to interactions with phonons (quantized lattice vibrations), it is called generation-recombination noise. Johnson noise is predominant at high frequencies, shot noise predominates at low frequencies, and

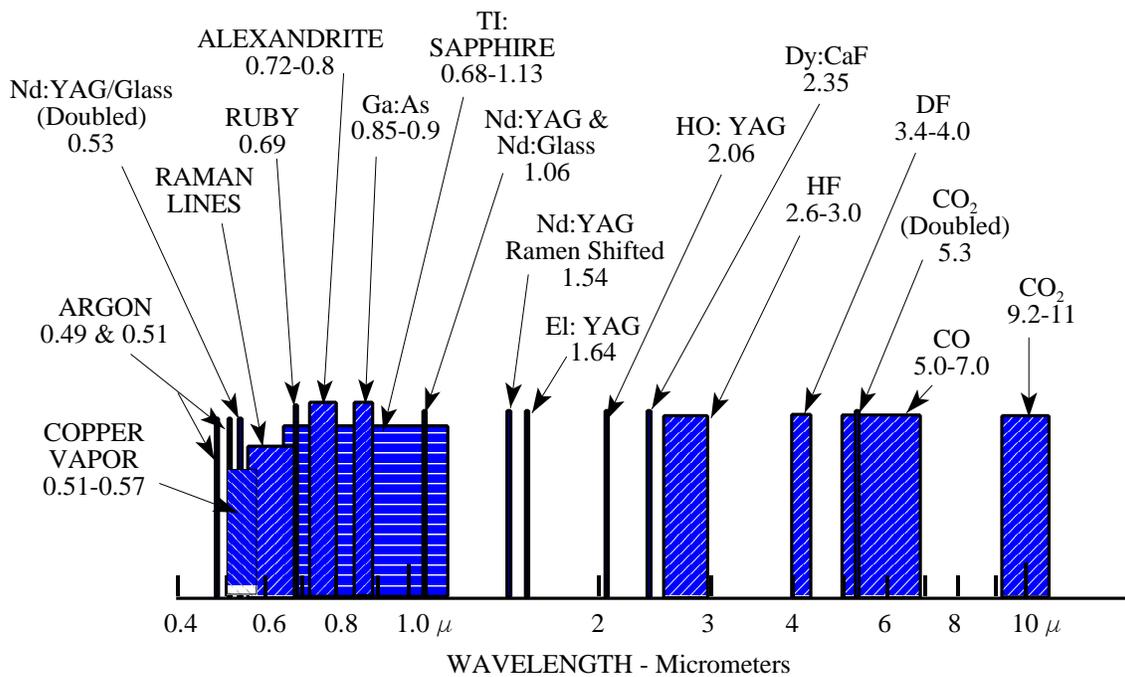
generation-recombination and photon noise are predominant at intermediate frequencies. As the name implies, 1/f noise has a power spectrum which is inversely proportional to frequency. It is dominant at very low frequencies. In photoemissive detectors it is called flicker noise and has been attributed to variation in the emission from patches of the photocathode surface due to variation in the work function of the surface. In semiconductors 1/f noise is also called modulation noise. Here it is apparently due to surface imperfections and ohmic contacts (which are a form of surface imperfection).

## LASERS

The word laser comes from Light Amplification by Stimulated Emission of Radiation. The lasing medium may be a solid, a gas, or a liquid. Lasing action has been achieved using atoms, ions, and molecules. The emission may be pulsed or CW.

Figure 19 shows the spectral output of several laser types.

The first laser was a pulsed, solid state laser, the ruby laser. In the ruby laser a xenon flash lamp is used to excite the atoms in a ruby rod to higher energy levels. The highly polished and mirrored ends of the rod form a resonant cavity. One end of the rod has a slightly lower reflectivity. The lamp excitation produces an inverted population of excited atoms which are stimulated to relax to lower energy levels releasing their extra energy as photons. Repeated reflections off the mirrored ends of the rod causes the photons to bounce back and forth through the rod stimulating further emissions at the same wavelength and phase producing a highly coherent beam which finally passes through the lower reflectivity end.



**Figure 19.** Spectral Lines / Ranges of Available Lasers

Figure 20 is a schematic representation of a ruby laser. The typical laser rangefinder uses a solid state laser with a neodymium-YAG crystal lasing at  $1.06 \mu\text{m}$ .

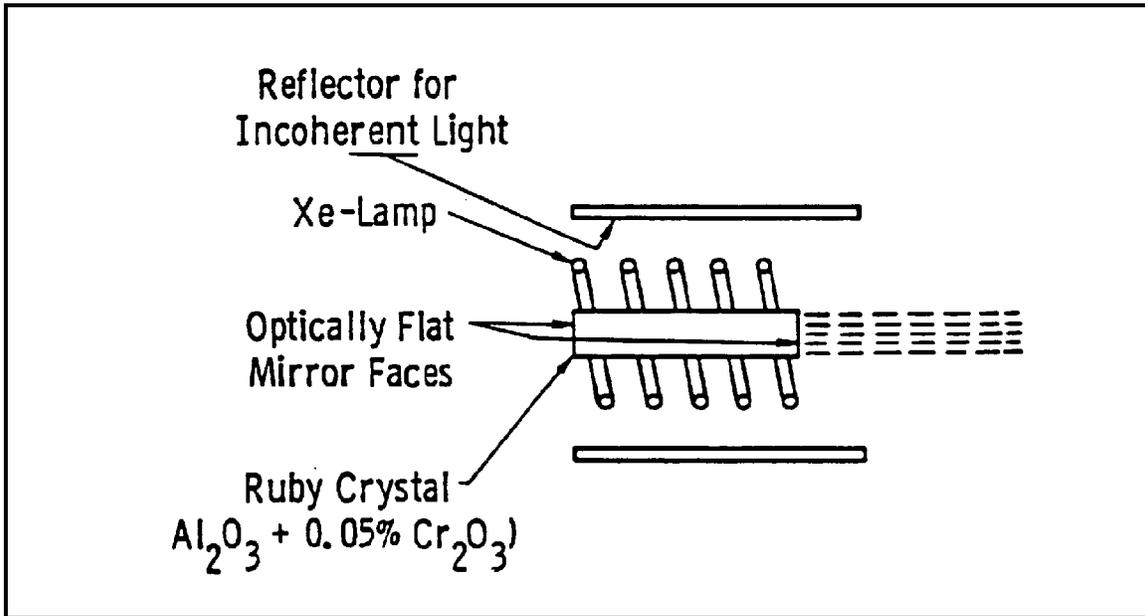


Figure 20. Ruby Laser

Gas lasers are of several kinds and can be pulsed or CW. The gas dynamic laser obtains its inverted population through a rapid temperature rise produced by accelerating the gas through a supersonic nozzle. In chemical lasers the inversion is produced by a chemical reaction. In the electric discharge laser the lasing medium is electrically pumped. The gas can also be optically pumped. In an optically pumped gas laser the lasing medium is contained in a transparent cylinder. The cylinder is in a resonant cavity formed by two highly reflective mirrors. The typical configuration is shown in Figure 21.

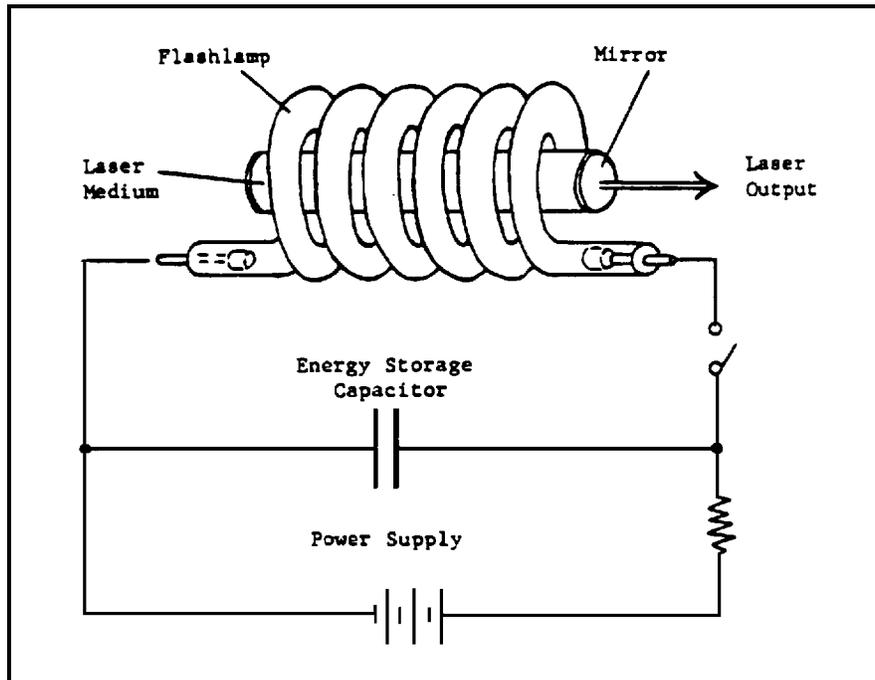


Figure 21. Gas Laser

Many gas lasers use carbon dioxide as the lasing medium (actually a mixture of CO<sub>2</sub> and other gases). These are the basis for most high energy or high power lasers. The first gas laser was an optically pumped CW helium-neon laser. The common laser pointer is a helium-neon laser operating at 0.6328 μm. The lasing medium is a mixture of helium and neon gas in a gas discharge or plasma tube as shown in Figure 22.

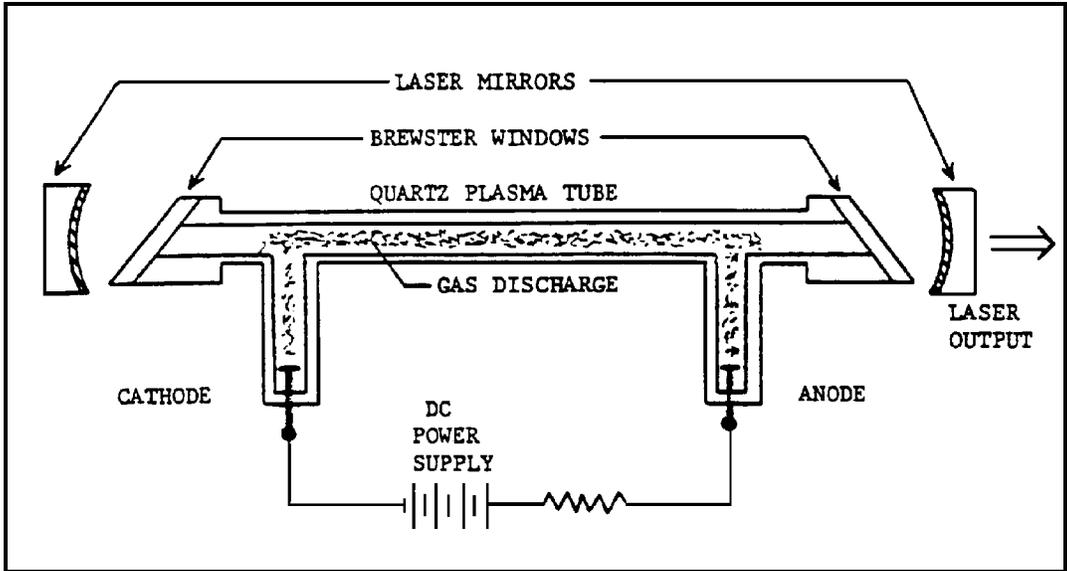


Figure 22. Helium-Neon Laser

The dye laser is an example of a laser using a liquid for the lasing medium. The lasing medium is an organic dye dissolved in a solvent such as ethyl alcohol. Dye lasers operate from the near UV to the near IR, are optically pumped, and are tunable over a fairly wide wavelength range.

Mention should also be made of semiconductor or injection lasers, also known as laser diodes. The junctions of most semiconductor diodes will emit some radiation if the devices are forward biased. This radiation is the result of energy released when electrons and holes recombine in the junction. There are two kinds of semiconductor diode emitters: (1) the light emitting diode (LED), which produces incoherent spontaneous emission when forward biased and which has a broad (800 angstrom) spectral output, and (2) the laser diode, which maintains a coherent emission when pulsed beyond a threshold current and which has a narrow spectral width (< 10 angstrom). In the laser diode the end faces of the junction region are polished to form mirror surfaces. They can operate CW at room temperatures, but pulsed operation is more common. Figure 23 shows a typical diode laser structure.

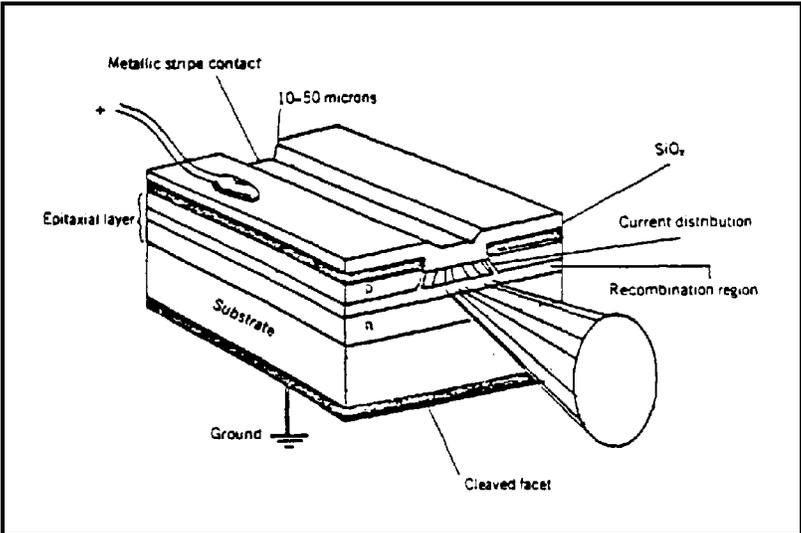
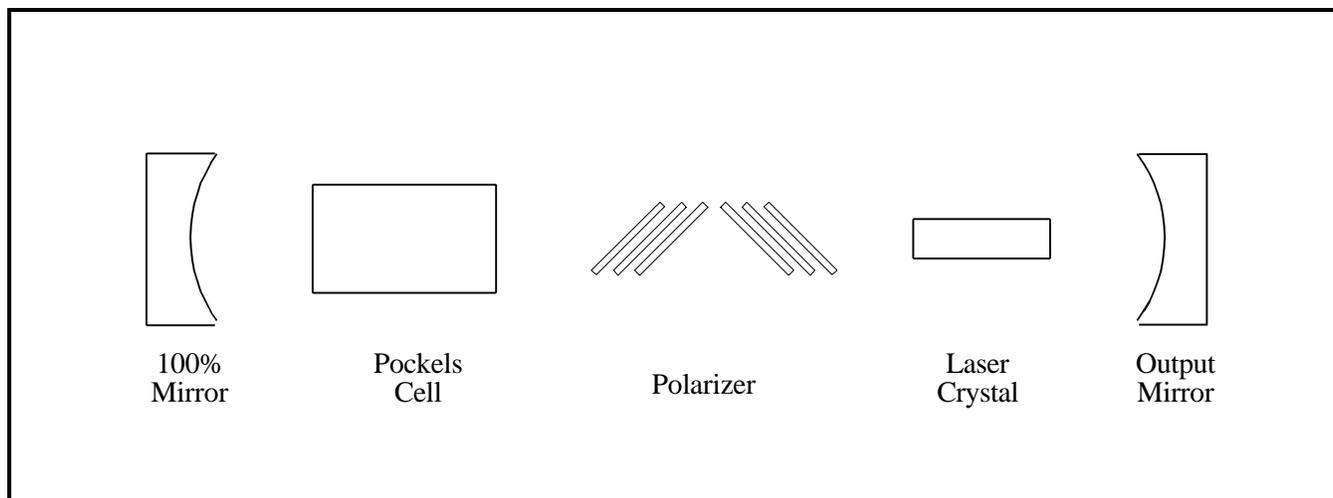


Figure 23. Diode Laser

Q-switching is a means of obtaining short intense pulses from lasers. The Q-switch inhibits lasing until a very large inverted population builds up. The switch can be active or passive. A passive Q-switch switches at a predetermined level. An active Q-switch is controlled by external timing circuits or mechanical motion. The switch is placed between the rod (or lasing medium) and the 100 percent mirror. Figure 24 shows an arrangement using a Pockels cell as an active Q-switch.



**Figure 24.** Q-switch Arrangement

## FIBER OPTICS

Fiber optic cables are the optical analogue of RF waveguides. Transmission of radiation through an optical fiber is due to total internal reflection of the radiation from the walls of the fiber. A plain fiber has leakage through the walls. This is controlled by coating, or cladding, the fiber with a lower refractive index material. Fibers with the best transmission characteristics (lowest attenuation) operate in the near infrared (out to  $1.7 \mu\text{m}$ ). Typical attenuations vary from two to ten dB/km in the visible to 0.2 to 0.5 dB/km in the near infrared. Developmental fibers for use in the 2 to  $20 \mu\text{m}$  wavelength range have attenuations of hundreds of dBs/km.

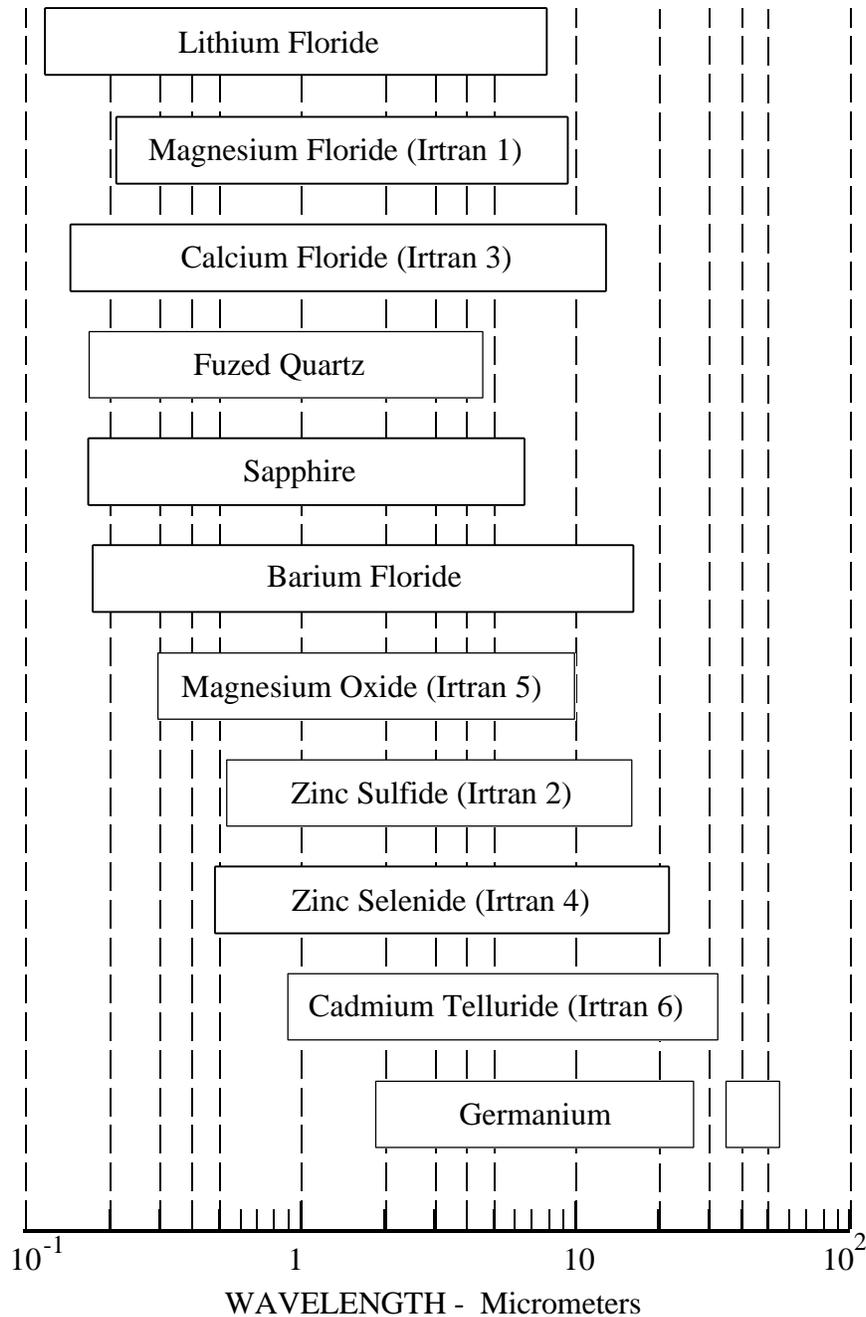
Optical fibers are not used in any current EO systems. Potential applications include use with smart skins where radiation is collected on the skin and piped by fiber optics to detectors elsewhere in the aircraft. Use of fiber optics in a high speed data bus for EW systems will probably come first.

## ELECTRO-OPTICAL SYSTEMS

A basic EO system is composed of an optical head, an electronics package, and an output unit. The optical head consists of a window, collecting optics which gathers the incident radiation and focusses it on the detector, a field stop to define the field of view, a reticle or chopper to modulate and encode the radiation, optical filters to define the wavelength region of response, a detector to convert the incident radiation into an electrical signal, and a preamplifier to increase the signal level from the detector before further handling or processing. The system electronics consist of amplifiers, signal processors, and system controls. The output unit consists of indicators or displays.

## Windows/Domes

For most applications of EO systems in EW the detection system is protected from the environment by a window or dome of optically transmissive material. The window operates both as a weather seal and, in some cases, helps to define the spectral response region of the system. The transmission bands of a representative sample of window materials is shown in Figure 25. The end points given are for the 10 percent transmission wavelengths. Not shown in Figure 25 are the various UV transmissive glasses such as Pyrex, Corex, and Vycor.



**Figure 25.** Transmission of Selected Window Materials

## Optical Filters

Most optical radiation detectors have a wider sensitivity band than desired for the particular application. To further define the system sensitivity, band interference filters or absorption filters are used. An absorption filter is a bulk material with a sharp cut-on or cut-off in its transmission characteristic. A cut-on and a cut-off filter can be combined to make a bandpass filter. By selecting absorption characteristics of absorption filters combined with the response of a detector, the desired system response can be obtained. An interference filter is composed of dielectric coatings on an appropriate substrate combined in such a way to produce cut-on, cut-off, or bandpass filters. Interference filters allow more control of the final response characteristics and smaller elements.

Besides bandpass filters, EO system optics often have antireflection (or AR) coatings to eliminate or greatly reduce unwanted reflections between optical elements.

## Detector Coolers

Many IR detectors have to be cooled for proper operation. Most systems use closed-cycle coolers or thermoelectric coolers. Thermoelectric coolers use the Peltier effect, which produces a reduced temperature by passing a d-c current through a thermoelectric junction. Multi-stage coolers can cool a detector down to below 200°K. Closed-cycle coolers typically are of the Stirling cycle design and utilize the expansion of a gas (helium) to cool a cold finger attached to the detector. These generally operate at liquid nitrogen temperature (77°K).

## Displays

Imaging systems such as Forward Looking Infrared (FLIR) systems use cathode ray tubes (CRTs) to display their output. Future EW systems may incorporate flat panel displays of some type. Possible types are liquid crystal displays (LCDs), LED arrays, or gas plasma displays.

## Types of Systems

EO systems of interest to EW include the following:

**FLIR systems** - A passive thermal imager which typically uses the emitted radiation of a target in the 8 to 14  $\mu\text{m}$  atmospheric window to produce a picture of the scene. Figure 26 shows the configuration of a typical FLIR using the serial scan approach. A FLIR could be used with a 10.6  $\mu\text{m}$  laser target designator to determine if the proper target is being illuminated.

**Infrared Search and Track Systems (IRSTS)** - The IRSTS is an EO analogue of a radar system. A focal plane array detector is scanned across the field of regard, and the locations of detected targets

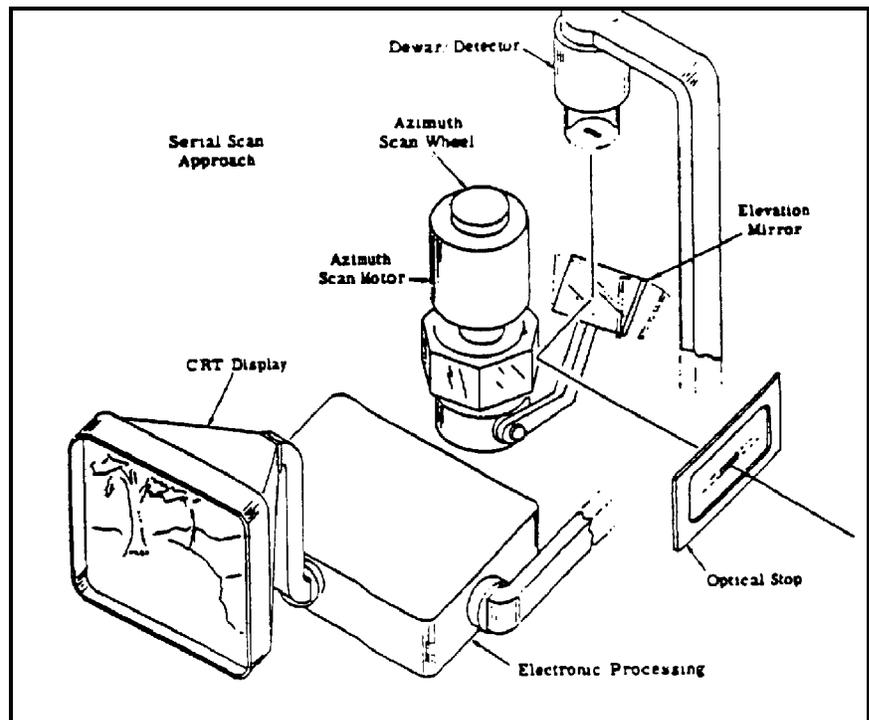


Figure 26. Serial Scan FLIR

are displayed on a CRT. Although without direct range measuring capability, triangulation techniques can be used for passive ranging. If combined with a laser rangefinder, an IRSTS could function just like an optical radar. An IRST provides better angular resolution but poorer range accuracy than a RF radar system.

Missile Warning Receivers/Sets - These may have either scanning or staring optical systems to detect and process the radiation from missile motors and alert the pilot that the aircraft is under attack.

Laser Warning Sets - These typically have staring optics. They detect and process received laser radiation. The pilot is alerted of the type and the direction of the laser detected.

Infrared Countermeasure (IRCM) Systems - The EO analogue of RF jammers. They radiate a modulated IR signal designed to confuse the detection/tracking system of an attacking IR guided missile and cause it to miss.

Television Camera Sets - High resolution TV camera systems primarily used for the identification friend or foe application.

Laser Rangefinders - A laser coupled with timing circuits to measure time of travel of laser pulses to and from a target. They can give very accurate ranges.

Laser Target Designators - Laser systems used to illuminate targets being attacked by laser guided munitions.